Angular Effects in Proton-Induced Single-Event Upsets in Silicon-on-Sapphire and Silicon-on-Insulator Devices

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Abstract—We present new data in the ongoing effort to bound the effect of proton angle of incidence on the single-event upset (SEU) rate in silicon-on-sapphire (SOS) and silicon-on-insulator (SOI) devices.

I. INTRODUCTION

Reed, et al. first predicted [1], [2] and later measured [3] how proton angle of incidence affects the measured single-event upset (SEU) cross-section in silicon-on-sapphire (SOS) and silicon-on-insulator (SOI) devices. This effect manifests itself as an increase in measured SEU cross-section at "glancing" incidence angles. We present the data on this effect for three additional SOS/SOI device types. Two of the device types presented here also give us a first look at the effect of radiation hardening by design (RHBD) on SEU sensitivity in SOS and SOI devices. The primary concern behind this continuing research is that failure of standard error prediction methods to consider this effective increase in the total cross-section, as a function of angle of incidence, will result in underprediction of error rates by up to an order of magnitude.

II. DEVICES TESTED

The Peregrine PE926C31 and PE926C32 RS-422 line driver/receiver pair were manufactured using their Ultra Thin Silicon (UTSi™) 0.5µm CMOS-on-sapphire process. Both the Honeywell HX6228 128k x 8 SRAM and the radiation-hardened reprogrammable field programmable gate array (RHrFPGA) were manufactured on the RICMOS™ IV SOI 0.7µm process.

<table>
<thead>
<tr>
<th>Part Number</th>
<th>Manufacturer</th>
<th>Function</th>
<th>Design Library</th>
</tr>
</thead>
<tbody>
<tr>
<td>PE926C31</td>
<td>Peregrine</td>
<td>RS-422</td>
<td>RHBD</td>
</tr>
<tr>
<td>PE926C32</td>
<td>Peregrine</td>
<td>Driver/Receiver</td>
<td>RHBD</td>
</tr>
<tr>
<td>HX6228</td>
<td>Honeywell</td>
<td>SRAM</td>
<td>Standard Honeywell RHBD</td>
</tr>
<tr>
<td>RHrFPGA</td>
<td>Honeywell</td>
<td>FPGA</td>
<td>Improved Honeywell RHBD</td>
</tr>
</tbody>
</table>

This work was supported by the NASA Electronic Parts and Packaging (NEPP) Program, NASA Flight Projects, the Defense Threat Reduction Agency (DTRA) under IACRO 03-40351 and 04-40641, Honeywell SSEC, and Peregrine Semiconductor.
III. TEST FACILITIES

All devices were tested for proton-induced SEU at the Indiana University Cyclotron Facility (IUCF) with an incident 205MeV proton beam. In addition, the Peregrine driver/receiver and the Honeywell SRAM were tested for proton-induced SEU at the University of California, Davis (UCD) Crocker Nuclear Laboratory (CNL) with a 63MeV proton beam.

IV. ANGLE CONVENTIONS

Figure 1 shows the representation of the angles as they are used here. Note that beam angle and angle of incidence are used interchangeably.

V. DEVICE DESCRIPTION, TEST SETUP AND PROCEDURES

A. Peregrine PE926C31/PE926C32

The PE926C31 and PE926C32 RS-422 driver/receiver pair is RHBD. Ohmic body contacts at dense spacings suppress any secondary charge collection that might occur. The receiver has two redundant comparators and a digital voter, unconnected in the data path, in the event that unacceptable SET performance had been seen from the analog circuitry. Parasitic device capacitance is minimized, and the devices also oppose drive strength in the digital section.

For the test of the Peregrine RS-422 driver/receiver, a 20 MHz square wave input signal (clock) was run through a divide-by-2 prior to the device under test (DUT) for timing purposes. Delay Net 1 (see figure 2) allowed us to line up the DUT output with the reference signal in the comparator. Delay Net 2 allowed us to line up the clock in the middle of each state in the square wave (the signal in the DUT is sampled on the rising edge of the clock signal, which is in the middle of the cycle in the DUT because of the divide-by-2). The signal was sampled at the midpoint of each high or low signal to determine if it had changed state; this prevents false errors due to rising/falling edge timing errors. The error counter tallies non-compares. The functionality of the setup was verified by deliberately altering one of the Delay Nets or removing power from the DUT to induce errors.

B. Honeywell HX6228 SRAM

The HX6228 is a rad-hard (1Mrad(SiO$_2$)) 1Mb SRAM designed for military applications and the space radiation environment. Typical read/write cycle times are 16 ns or less, and 25 ns or less across the full military temperature range. The RICMOS™ IV process is a 5V, SIMOX CMOS technology with a 150Å gate oxide and a minimum feature size of 0.7µm (0.55µm effective gate length). A seven-transistor memory cell is used for single-event upset hardening, while three-layer metal power bussing and the low collection volume SIMOX substrate provide dose rate hardening.

In the Honeywell SRAM, a pattern of alternating 0’s and 1’s was written to the device prior to irradiation. The address counter initially cycled through each memory location while the predetermined data pattern is written. During irradiation, the data were read and rewritten continuously and then compared to a reference buffer. Once the comparison was completed, SEUs (incorrect data values) were written to FIFO.
1 and the address location was captured and written to FIFO 2 when a miss-compare is detected. Data and address information are then downloaded to a computer for analysis. A block diagram of the test setup is given in figure 3.

![Block Diagram of Honeywell HX6228 SRAM Test Setup](figure3.png)

Figure 3. Diagram of Honeywell HX6228 SRAM test setup.

C. Honeywell RHrFPGA

The RHrFPGA is an SRAM-based (reconfigurable) field-programmable gate array manufactured by Honeywell using an RH SOI fabrication process. The RICMOS™ IV process is a 5V, SIMOX CMOS technology with a 150Å gate oxide and a minimum feature size of 0.7μm (0.55μm effective gate length). It has 6,400 user-configurable logic cells and 131,152 configuration SRAM cells. NASA GSFC funded the development, fabrication, and radiation testing of the RHrFPGA.

Proton testing was performed at IUCF on the RHrFPGA because it contains SRAM memory elements that were hardened after an earlier Honeywell SRAM that exhibited susceptibility to upset depending on proton angle of incidence. This sensitivity was attributed to a single secondary heavy ion hitting two transistors within a memory cell. The cross-section would be highest if the incident protons were parallel to the path between two sensitive transistors in a cell.

At IUCF, all irradiations were carried out at a 70° angle of incidence at roll = 0° and roll = 90° because the configuration RAM and the application flip-flops are orthogonal to each other. The test was limited to 70° because of constraints in rotating the fixture and concerns about irradiating the control device. A block diagram of the test setup is given in figure 4.

![Block Diagram of Honeywell RHrFPGA Proton Test Setup at IUCF](figure4.png)

Figure 4. Diagram of Honeywell RHrFPGA proton test setup at IUCF.

VI. TEST RESULTS

A. Peregrine PE926C31/PE926C32

At IUCF, the devices were exposed to a 205MeV proton beam at a 90° angle of incidence (grazing angle). The driver was only exposed at a roll of 0° and the receiver was exposed at roll = 0° and roll = 90°. No single-event transients (SETs) were observed to a fluence of 3.4 x 10¹² p/cm² for each device type at IUCF.

At UCD, the devices were exposed to a 63MeV proton beam at roll = 0° and roll = 90° at a number of angle of incidence between 0° and just over 90°. The results at UCD presented a very different picture. For 0° roll, there was only one SEU captured with the beam perpendicular to the die and again only one SEU with the beam grazing the surface of the die in this orientation. In the 90° roll condition, there was more than an order of magnitude difference in the cross section compared to the 0° roll (see figure 8). These data would imply an interesting geometry in the sensitive volume that should be studied in more detail, including circuit simulation.

B. Honeywell HX6228 SRAM

At IUCF, the devices were exposed to a 205MeV proton beam at roll = 0° and roll = 90° at a number of angles of incidence between 0° and just over 90°. The results showed an angle of incidence effect on the order of a factor of 2 increase in proton cross-section with angle. There was little difference in the results for either roll direction (see figures 5-7). The cross sections given are in cm²/device.

At UCD, the devices were exposed to a 63MeV proton beam at roll = 0° and roll = 90° at a number of angle of incidence between 0° and just over 90°. The results at UCD present a very different picture. For 0° roll, there was only one SEU captured with the beam perpendicular to the die and again only one SEU with the beam grazing the surface of the die in this orientation. In the 90° roll condition, there was more than an order of magnitude difference in the cross section compared to the 0° roll (see figure 8). These data would imply an interesting geometry in the sensitive volume that should be studied in more detail, including circuit simulation.
C. Honeywell RHrFPGA

At IUCF, the devices were exposed to a 205MeV proton beam. All exposures were performed at a 70° angle of incidence in the roll = 0° and roll = 90° test configurations. No SEUs were observed to a fluence of 3.4 x 10^{13}p/cm^2 in both test configurations.

VII. COMPARISON TO PREVIOUS DATA

In 2002 Reed, et al [3] gathered data on devices manufactured on similar processes to those presented here. All of the devices tested then were exposed in the roll=90° condition. The Peregrine PE9301 prescaler was exposed to 63MeV protons at UCD and 200MeV protons at IUCF. The peak to normal incidence (PTNI) ratio for 200MeV protons was 1 (no change in cross section with angle of incidence) and the PTNI ratio for 63MeV protons was ~12. The Honeywell HX6408 4M SRAM was exposed to 158MeV protons at IUCF. The PTNI ratio was ~10. However, data was only taken up to an angle of 73°.

The Honeywell HX6228 gave interesting results shown below in Table 2.

<table>
<thead>
<tr>
<th>TABLE II. PTNI RATIO S FOR THE HONEYWELL HX6228</th>
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<tbody>
<tr>
<td>Roll (°)</td>
</tr>
<tr>
<td>---------</td>
</tr>
<tr>
<td>0</td>
</tr>
<tr>
<td>90</td>
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</tbody>
</table>

*The cross section used for normal incidence was 1x10^{-19}cm^2/bit, however, there was no SET recorded for normal incidence in this test configuration.
VIII. DISCUSSION

Our results show that RHBD best practices can significantly decrease SEU sensitivity of SOS and SOI devices, in some cases eliminating proton-induced SEU sensitivity. The results also indicate that for some devices, the roll of the DUT can impact proton sensitivity and may have implications for heavy-ion irradiation as well. In some cases, this roll effect can be greater in magnitude than the previously observed enhancements of proton SEU cross-section seen for protons at grazing angle of incidence. These results have significant implications for both testing and rate prediction.

It should be noted that the Peregrine devices tested were only operated at 20 MHz. The potential for increased SEU sensitivity at much higher speeds needs to be examined.

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